§ 25.1585

described in terms of accelerations, must be furnished.

[Doc. No. 5066, 29 FR 1891, Dec. 24, 1964, as amended by Amdt. 25–38, 41 FR 55468, Dec, 20, 1976; Amdt. 25–42, 43 FR 2323, Jan. 16, 1978; Amdt. 25–46, 43 FR 50598, Oct. 30, 1978; Amdt. 25–72, 55 FR 29787, July 20, 1990; Amdt. 25–105, 66 FR 34024, June 26, 2001]

§25.1585 Operating procedures.

- (a) Operating procedures must be furnished for—
- (1) Normal procedures peculiar to the particular type or model encountered in connection with routine operations;
- (2) Non-normal procedures for malfunction cases and failure conditions involving the use of special systems or the alternative use of regular systems; and
- (3) Emergency procedures for foreseeable but unusual situations in which immediate and precise action by the crew may be expected to substantially reduce the risk of catastrophe.
- (b) Information or procedures not directly related to airworthiness or not under the control of the crew, must not be included, nor must any procedure that is accepted as basic airmanship.
- (c) Information identifying each operating condition in which the fuel system independence prescribed in §25.953 is necessary for safety must be furnished, together with instructions for placing the fuel system in a configuration used to show compliance with that section.
- (d) The buffet onset envelopes, determined under §25.251 must be furnished. The buffet onset envelopes presented may reflect the center of gravity at which the airplane is normally loaded during cruise if corrections for the effect of different center of gravity locations are furnished.
- (e) Information must be furnished that indicates that when the fuel quantity indicator reads "zero" in level flight, any fuel remaining in the fuel tank cannot be used safely in flight.
- (f) Information on the total quantity of usable fuel for each fuel tank must be furnished.

[Doc. No. 2000-8511, 66 FR 34024, June 26, 2001]

§25.1587 Performance information.

- (a) Each Airplane Flight Manual must contain information to permit conversion of the indicated temperature to free air temperature if other than a free air temperature indicator is used to comply with the requirements of §25.1303(a)(1).
- (b) Each Airplane Flight Manual must contain the performance information computed under the applicable provisions of this part (including §§ 25.115, 25.123, and 25.125 for the weights, altitudes, temperatures, wind components, and runway gradients, as applicable) within the operational limits of the airplane, and must contain the following:
- (1) In each case, the conditions of power, configuration, and speeds, and the procedures for handling the airplane and any system having a significant effect on the performance information.
- (2) V_{SR} determined in accordance with §25.103.
- (3) The following performance information (determined by extrapolation and computed for the range of weights between the maximum landing weight and the maximum takeoff weight):
- (i) Climb in the landing configura-
- (ii) Climb in the approach configuration.
 - (iii) Landing distance.
- (4) Procedures established under $\S25.101(f)$ and (g) that are related to the limitations and information required by $\S25.1533$ and by this paragraph (b) in the form of guidance material, including any relevant limitations or information.
- (5) An explanation of significant or unusual flight or ground handling characteristics of the airplane.
- (6) Corrections to indicated values of airspeed, altitude, and outside air temperature.
- (7) An explanation of operational landing runway length factors included in the presentation of the landing distance, if appropriate.

[Doc. No. 2000–8511, 66 FR 34024, June 26, 2001 as amended by Amdt. 25–108, 67 FR 70828, Nov. 26, 2002